

# Finite element analysis for airfield asphalt pavements rutting prediction

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**Abstract.** This paper presents a numerical study of an aircraft wheel impacting on a flexible pavement. The proposed three dimensional model simulates the behaviour of flexible runway pavement during the landing phase. This model was implemented in a finite element code in order to investigate the impact of repeated cycles of loads on pavement permanent deformation.

In the model a traditional multi-layer pavement structure was considered. In addition, the asphalt layer (HMA) was assumed to follow an elasto-viscoplastic behaviour.

The results demonstrate the capability of the model in predicting the permanent deformation distribution in the asphalt layer.

**Key words:** asphalts, FEM; airfield pavement; aircraft landing, viscoelasticity, creep.

## 1. Introduction

In recent years several studies have analysed the behaviour of flexible pavements developing 3D finite element models that are capable of accurately determining stresses and deformation in pavements caused by aircraft with multiple-wheel landing gear configuration.

Flexible pavements can often be idealized as closed system consisting of several linear elastic layers, with each layer both uniform in thickness and infinite in horizontal extent. This simplified approach to pavement modelling is no longer acceptable [1]. Differently from the layered theory, the FE method can be a complex and costly analysis tool; it is thus employed only when accurate results are needed. Although involving a more complicated formulation than the multi-layer elastic theory, the application of FE techniques is generally thought to allow an accurate simulation of pavement problems. Furthermore this method allows to consider almost all controlling parameters (e.g., dynamic loading, discontinuities such as cracks and shoulder joints, viscoelastic and nonlinear elastic behaviour, infinite and stiff foundations, system damping, quasi-static analysis, crack propagation) [1].

For example, the model used in the study conducted by Zaghoul and White [2] employed 3D dynamic finite element to investigate the response of moving loads on pavement structures. Zaghoul employs a visco-elastic model for the asphalt concrete, an extended Drucker-Prager model for granular base course and Calm Clay model for the clay subgrade soils [3].

Taciroglu [4] simulated the pavement responses using three-dimensional finite element analysis and adopted the  $K-\theta$  model and the Uzan model as the nonlinear unbound granular material model and linear subgrade soils model.

Kim [5] found that nonlinearity of unbound layers using the Drucker-Prager plasticity model was not suitable to

pavement analyses. Therefore, the Uzan model was adopted for granular materials and cohesive soils for the nonlinear analysis. Mohr-Coulomb failure criterion was employed in the nonlinear finite element analysis.

Erlingsson [6] conducted three-dimensional finite element analyses of a heavy vehicle simulator used to test low volume road structures. A linear elastic material model was used and the single and dual wheel configurations were given.

In recent years, researchers have successfully applied linear viscoelastic theory to describe the behaviour of HMA materials. Elseifi et al. [7] conducted a comparative study between the elastic FE model and the linear viscoelastic FE model and concluded that it is imperative to incorporate a viscoelastic constitutive model into pavement design methods for improved accuracy. Yin et al. [8] showed that 3-D finite element modelling utilizing viscoelastic material properties provides reasonable prediction of strain response in the field. However, HMA only behaves elastically at low temperature and a linear stress-strain relationship is incapable of predicting the nonlinear creep responses to vehicular loads. Time-, temperature- and stress-dependent nonlinear behaviour of HMA resulting in permanent deformation may only be described by creep deformation or plasticity.

A constitutive model for describing the plastic behaviour of HMA is difficult to find, even if an extensive laboratory testing program is performed. Chehab [9] developed an advanced material characterization procedure including the theoretical models and its supporting experimental testing protocols necessary for predicting the response of asphaltic mixtures subjected to tension loading. The model encompasses the elastic, viscoelastic, plastic and visco-plastic components of asphalt concrete behaviour.

Onyango [10] evaluated existing mechanistic models that predict permanent deformation (rutting) in asphalt mixes by

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comparing computed permanent deformation to that measured in a full-scale accelerated pavement test. To model the permanent deformation of HMA, it is important to utilize a constitutive model that can account for cumulative creep strains. Nonlinear creep models can be used to describe material response similar to that of a stress dependent plasticity model.

Pirabarooban et al. [11] successfully developed a elasto-viscoplastic creep model representing the time-dependency of asphalt mixtures to evaluate their rutting potential and to identify factors having a significant effect. The creep model parameters were derived from Asphalt Pavement Analyzer test results.

Huand et al. [12] conducted 3D simulations of asphalt pavement sections using a nonlinear viscoelastic and viscoplastic model. The results of this research show that the capability to capture the pavement responses under different temperatures.

Huang [13] proposed a total cumulative time loading approach to simulate a large number of loading cycles of moving loads in 3D FE simulation. Hua [14] and Al-Qadi et al. [15] improved this cumulative loading time approach. In particular, Al-Qadi et al. [15] and Fang et al. [16] illustrated the effectiveness of using a nonlinear time-hardening creep model to compute permanent deformation

In order to evaluate creep strain of HMA in flexible pavements, the proposed 3-D FE model implements a nonlinear time-hardening creep model that describes the behaviour of HMA in the primary and secondary stages. The principal aim is to assess the effects of repeated heavy impacts caused by aircraft landing gear wheels on the pavement. In particular, this study wants to simulate the pavement performance under aircraft landing in order to investigate the relationship between the rutting depth and the number of loads.

## 2. Materials and methods

**2.1. Pavement model.** Flexible pavements can often be idealized as closed systems consisting of several layers; so it was decided to model the surface, base, sub-base and sub-grade material using three dimensional finite elements.

The pavement structure in the application is based on the structure as found for the runway of the Reggio Calabria (Italy) airport, and it consists of a 100 mm thick asphalt concrete layer as the surfacing course, a 150 mm thick of bitumen-treated mixture as the base course, a 210 mm thick granular layer as the subbase course and a compacted soil subgrade (Fig. 5).

**2.2. Contact area and associated stress.** The most common way of applying wheel loads in a finite element analysis is to apply pressure loads to a circular or rectangular equivalent contact area with uniform tire pressure. Even such a simple model of this impact as an uniformly distributed load of over a circular or rectangular area is commonly used in research and computing programs [17–20]. The contact area can be calculated as:

$$A_c = \frac{P}{p}, \quad (1)$$

where  $P$  is the wheel load and  $p$  is the tire pressure. In the model the Airbus 321 tires were considered [21].

Table 1  
A321 Characteristics

Maximum ramp weight	83400 kg
Percentage of weight on main gear group	95.4%
Nose gear tire size	30×8.8 R15
Nose gear tire pressure	10.8 bar
Main gear tire size	1270×455 R22
Main gear tire pressure	13.6 bar

In the finite element model, the contact area,  $A_c$  was represented as a rectangle having a length  $L$  and a width  $L' = 0.7 \cdot L$ . To evaluate the pavement response in exceptional condition, the dynamic parameters of a “hard” landing (Fig. 1), were considered [22].

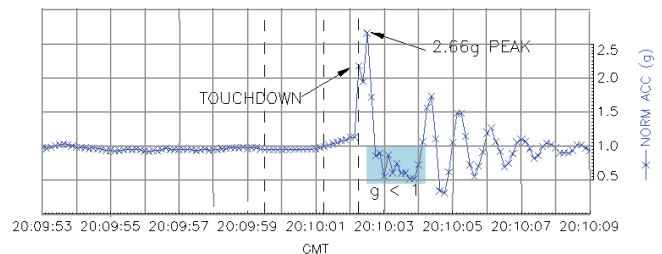


Fig. 1. Normal acceleration during landing phase after Ref. 22

Starting from this, considering the aircraft of mass  $m$  (ignoring gear mass) with the main gear shock absorbers represented by a linear model of total damping  $C_M$  and stiffness  $K_M$ . Thus a single DoF model is considered, where any fore-and-aft offset of the main gear from the aircraft centre of mass, and consequent pitch motion, has been ignored. The equation for the aircraft as the gear comes into contact with the ground as shown in Fig. 2 is then:

$$m\ddot{z}_M + C_M\dot{z}_M + K_M z_M = L - mg = 0, \quad (2)$$

where  $z_M$  is measured from the aircraft position with the leg uncompressed. The weight  $mg$  has been included as a steady force because the final solution (once wing lift  $L$  has reduced to zero) must show a steady leg deformation equal to the sag of the aircraft on its landing gear. However, with lift  $L$  present for the initial landing impact, sag will not occur since lift offsets the weight; the weight will only transfer on to the gear once the lift is ‘dumped’.

At the moment when the gear comes into contact with the ground, the aircraft is descending at a velocity  $W_E$  (vertical landing velocity); note that the aircraft is also moving forward but this effect is ignored when using such a simple model to show the energy dissipation in the vertical direction. The initial conditions at the moment of impact are then  $z_M(0) = 0$ ,  $\dot{z}_M(0) = W_E$ , which will lead to free vibration of the aircraft on its gear.

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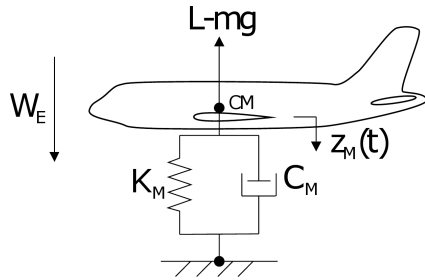


Fig. 2. Aircraft landing gear model (Ref. 23)

From (2) it is possible to calculate the acceleration graph during the hard landing (Fig. 3). As shown in Fig. 3 the peak deceleration value, during the hard landing, is about  $2\text{-}g$   $\text{m/s}^2$ . This value was used in the finite element model to calculate the maximum load of the main gear wheel. Under this load the contact area becomes:

$$A_c = \frac{397025}{1.36} = 291930 \text{ mm}^2. \quad (3)$$

Form (3) the footprint dimensions are:  $L = 646$  mm and  $L' = 442$  mm.

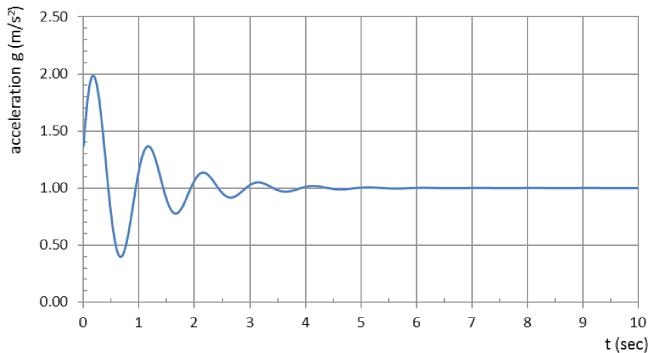


Fig. 3. Acceleration graph

**2.3. Materials proprieties.** The pavement configuration is shown in Fig. 5 and the material properties of pavement layers are given in Table 2.

Table 1  
Layers thickness and elastic material properties

Layer	Thickness [mm]	Modulus of elasticity [MPa]	Poisson's ratio
Surface	100	7000	0.30
Base	150	2000	0.35
Subbase	210	400	0.35
Subgrade	infinite	70	0.33

Elastic properties (modulus of elasticity and Poisson's ratio) were obtained by conducting laboratory testing on HMA materials and field non-destructive evaluation of granular and subgrade materials [24, 25].

For simplicity, the material proprieties of the granular layers and the base course were assumed to be time-independent and linear elastic.

The nonlinear time-hardening creep model has been used to simulate the behaviour of the HMA surface layer.

Unlike the linear viscoelastic model, the nonlinear time-hardening creep model can determine the accumulation of permanent deformation or rutting with time at different temperatures [14–16, 26].

Typical nonlinear creep behaviour can be divided into primary, secondary, and tertiary stages. The creep strain rate decays during the primary stage and constant creep strain rate is reached at the secondary stage. Creep fracture may occur at the end of the tertiary zone (Fig. 4).

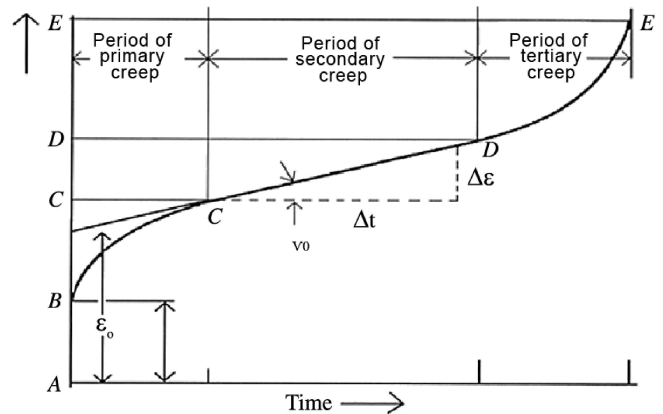


Fig. 4. Typical deformation characteristic of HMA under constant stress condition

The formulation is based on a basic assumption that material depends on the present stress state explicitly. In this approach, strain rate is represented by:

$$\dot{\epsilon}^c = \frac{\partial \epsilon^c}{\partial t} = A \sigma^n t^m, \quad (4)$$

where  $\epsilon$  – uniaxial equivalent creep strain rate,  $\sigma$  – uniaxial equivalent deviatoric stress (Misses equivalent stress),  $t$  – total time,  $A$ ,  $n$ ,  $m$  – user defined constants functions of temperature.  $A$  and  $n$  must be positive with  $-1 < m \leq 0$ .

Equation (4) is the time-hardening creep formulation, which shows that the creep strain rate depends on stress and time. In this research, the temperature was assumed to be constant ( $20^\circ\text{C}$ ) and the  $m$  value was set at  $-0.5$  while  $n$  value is  $0.67$  and  $A$  is  $1.0\text{e-}9$ .

**2.4. Finite element model.** The considered pavement section has the following dimensions: 5 m in  $x$  and  $y$  directions and 3.5 m in the  $z$ -direction. Figure 5 presents a sketch of the pavement structure geometry with the model characteristics.

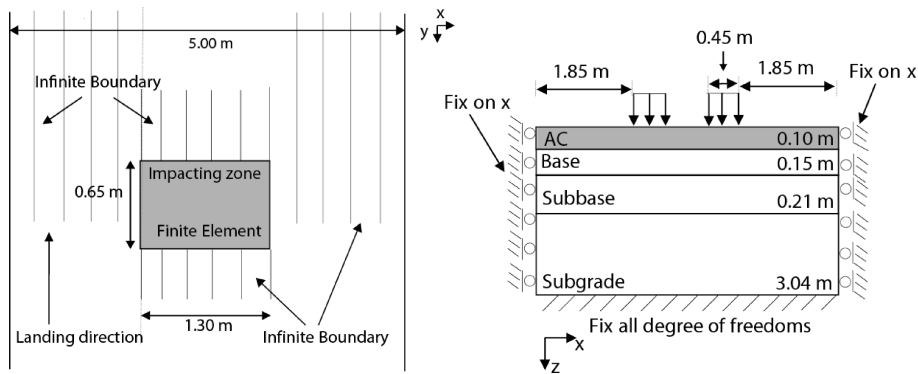


Fig. 5. Sketch of the simulated pavement structure

The load of two main gear wheels during the landing phase was assumed to be symmetrically applied on the pavement surface. The model was constrained at the bottom; along the sides parallel to  $y$ -axis all nodes were constrained horizontally but were free to move in vertical direction. The boundary nodes along the pavement edges parallel to  $x$ -axis were free in both horizontal and vertical directions [10, 27, 28]. Different FE analysis on model with different boundary condition at the edges parallel to  $x$  and  $y$  axis were performed to examine the boundary effect. No significant effect was found on surface deformations.

All layers were considered perfectly bonded to one another so that the nodes at the interface of two layers had the same displacements in all three ( $x, y, z$ ) directions. Assuming perfect bond at the layer interfaces implies that there will be no slippage at the interface. This assumption is more applicable to hot mix asphalt layers, since the possibility of slippage is greater at the subbase/subgrade interface [8].

The degree of mesh refinement is the most important factor in estimating an accurate stress field in the pavement: the finest mesh is required near the loads to capture the stress and strain gradients. The mesh presented has 69090 nodes and 64124 elements (49200 infinite linear hexahedral elements of type CIN3D8 and 14924 linear hexahedral elements of type C3D8R). The FE model employs the infinite element to represent the infinite boundary in the landing direction. The finite element mesh considered for the analysis is shown in Fig. 6.

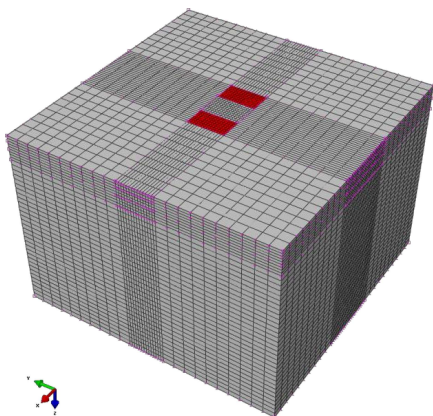


Fig. 6. Three-dimensional view of the finite element model showing the area of applied load

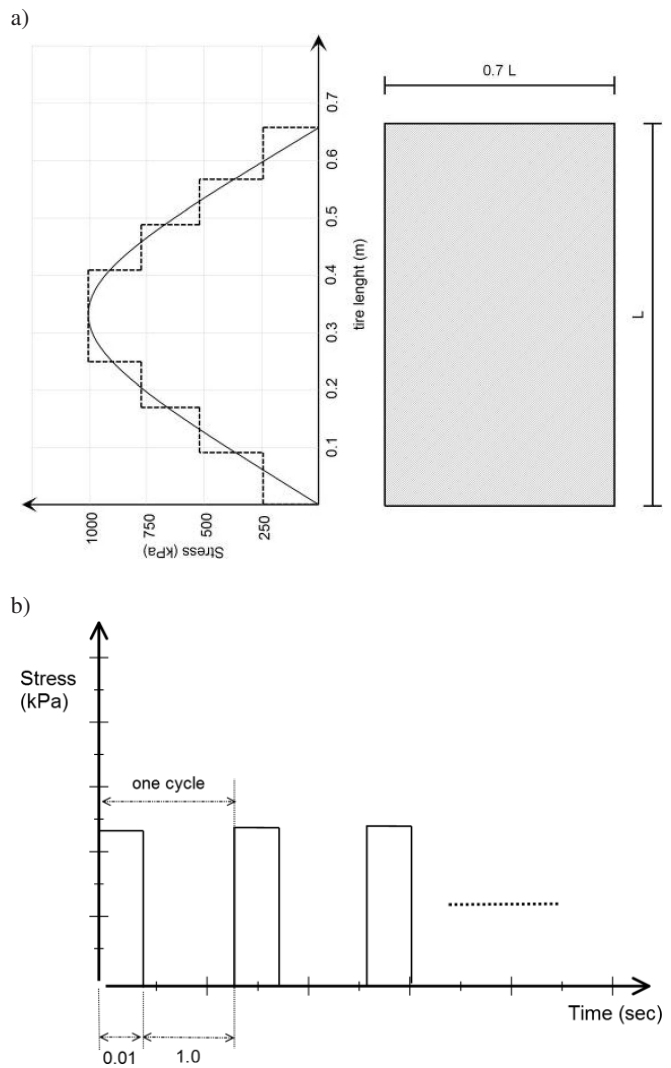


Fig. 7. Load pressure distribution over the tire contact (a) and schematic of applied cyclic load (b)

The loads were applied on the pavement surface with a maximum pressure of 1060 kPa and a distribution over the contact length [20] as illustrated in Fig. 8a. For simplicity, the dotted curve was used to represent the contact pressure distribution for each tire in the FE simulation.



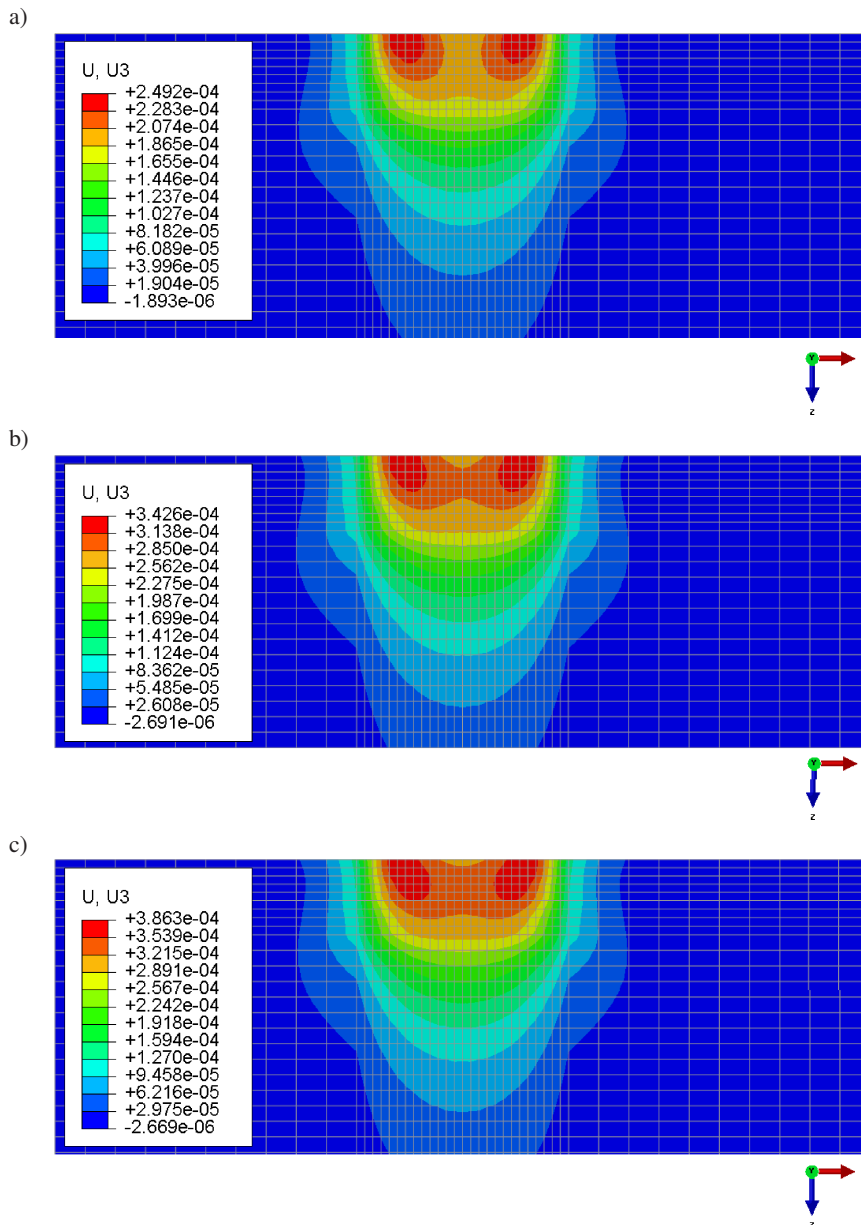


Fig. 8. Contours plot of the model displacement ( $z$ -direction) after a) 100, b) 500, c) 1000 cycles

Within each cycle, the load is applied with a duration time of 0.01 sec in order to simulate the aircraft landing speed.

To analyse the behaviour of the pavement structure under repeated load cycles, in the simulation the load is removed for 1.0 sec as shows in Fig. 8b.

### 3. Results and discussion

The results of the simulation are illustrated in the following figures. The displacement is considered as a response of applying repeated loads. The final vertical displacement  $U_z$  beneath the center of the load after 1000 cycles of loading is 0.39 mm (Fig. 8c).

A part of these displacements is recovered at the end of the load pulse, according to the resilient properties. The oth-

er part perpetuates. The permanent response is related to the plastic strains and represents the field rutting.

Figure 9 shows permanent deformation (rutting) across the transverse section under the center of loading for different cycles of load. The figure illustrates that the permanent deformation increases with increasing the number of loading cycles. This results demonstrates that the model is able to capture the pavement behavior under repeated loads.

Figure 10 shows the relationship between rutting depth (mm) and the number of load cycles ( $N$ ). This figure shows that the accumulation rate of rutting becomes smaller with an increase in loading cycles. Moreover, after  $N = 500$ , the asphalt material reaches the secondary creep stage.

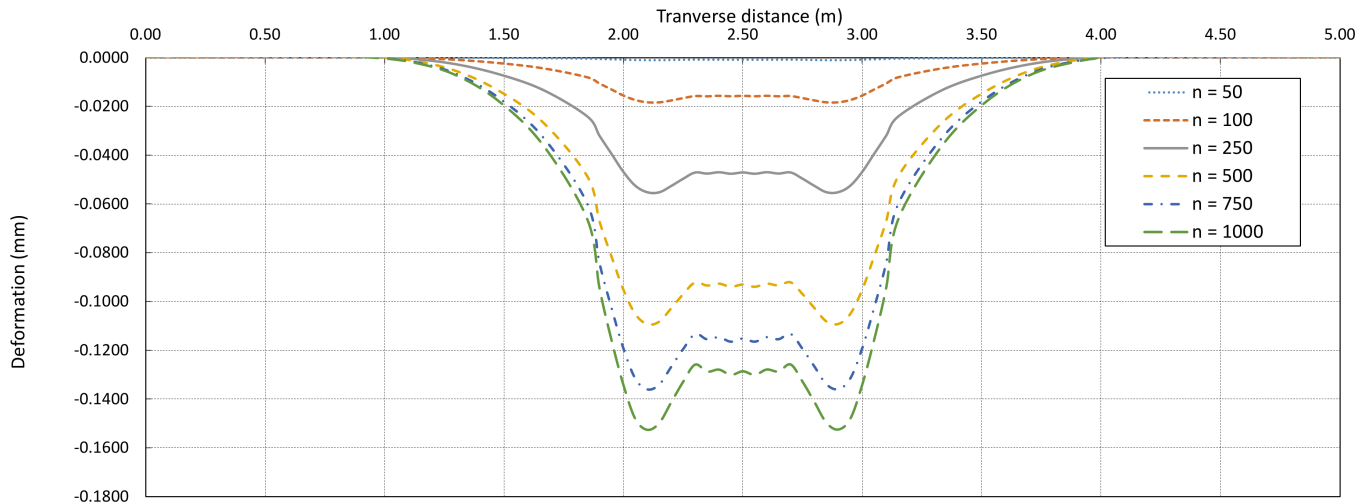


Fig. 9. Permanent deformation profile for different number of cycles

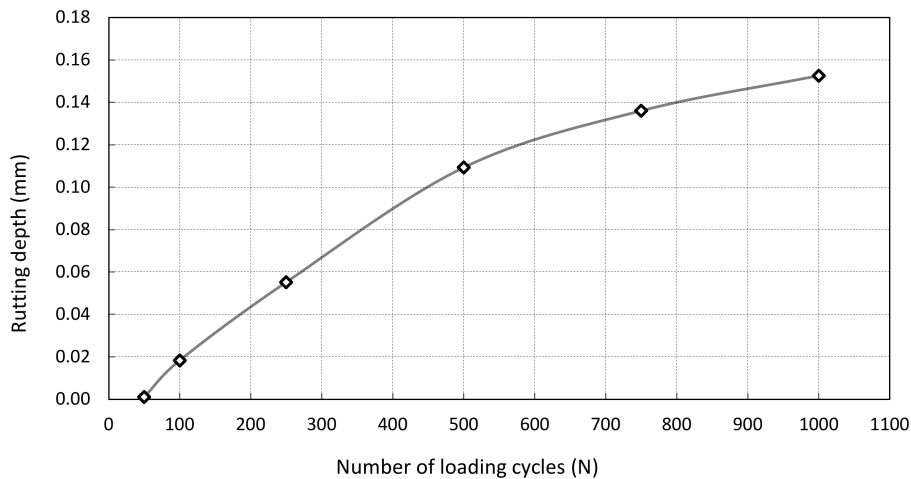


Fig. 10. Relationship between the number of cycles and the rutting depth

#### 4. Conclusions

The finite element analysis of pavement structures, if validated, can be extremely useful, because it can be used directly to estimate pavement response parameters without resorting to potentially costly field experiments.

If accurate correlations between the theoretically-calculated and the field-measured response parameters can be obtained, then the finite element model can be used to simulate pavement response utilizing measurements from strain gages. In particular, the proposed model has clearly confirmed the need and importance of 3-Dimensional finite element analyses on flexible pavements to consider the behaviour of the structure under high stress.

In particular, the importance of this research arises from the fact that several agencies adopted the rutting as failure criterion in pavement design. The simulation results from this study show that the proposed model has the capability to capture the pavement responses under repeated loads.

Future research advancements can be done in the direction of a better profiling of the pavement behaviour under stress, in function of different combinations of variables as temperature, tire type and pressure and the comparison with field pavement performance data will be conducted.

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*Finite element analysis for airfield asphalt pavements rutting prediction*

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